RUSSIAN AEROSPACE AGENCY Federal State Enterprise Experimental Design Bureau "Fakel"

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FINAL REPORT ON PROJECT #1590

"DEVELOPMENT OF SMALL SPT DEMO MODEL"

Project Duration: 01.11.1999 - 30.04.2000.

Project Manager:

Dr. Arkhipov Boris A. Tel. +7 0112 456600

<u>Collaborator:</u> Airforce Phyllips Laboratory, AFRL/PRRS 4 Draco Drive, CA,USA, 93523

<u>Partner:</u> European office of Aerospace Research and Development (EOARD) 223/231 Old Marylebone Road, London, UK, NW1 5TH

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4. TITLE AND SUBTITLE			5. FUN	DING NUMBERS
Development Of The Small S	PT Demo Model			ISTC Registration No: 1590
6. AUTHOR(S)		· · · · · · · · · · · · · · · · · · ·		
Dr. Boris Arkhipov				
7. PERFORMING ORGANIZATION NA	ME(S) AND ADDRESS(ES)		8 PER	FORMING ORGANIZATION
EDB Fakel.	ME(3) AND ADDICESS(ES)			ORT NUMBER
181 Moskovsky prospekt Kaliningrad (obl.) 236001 Russia				N/A
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13. ABSTRACT (Maximum 200 words)				
according to the project it is n SPT with thrust efficiency > maximising thrust @ 100 W ir To develop, manufacture and to provide integrated tests; □ deliver this unit to the tests are complete, the t □ Facilities/Equipment □ □ at Fakel's manufacturing capalength not less than 1.5 m for length not less than 1 m for thower than 3 x 10^-5 Torr by chamber.□□Schedule of Repplan. □□ Payment \$10,0000 Deliveries: Small 3□□ Small SPT unit tes	I test the cathode laboratory model for t -To prepare technical description an Edwards AFB for testing.□□Fakel rep hruster unit will be left at Edwards AF □To fulfil the tasks presented in Part B abilities. The test facilities are as follow the integrated thruster unit te ne autonomous cathode and N - Xe. □□The thruster will then be test norts/Deliveries□□Milestone 1□□ So □□□Milestone 2□□ Small SPT uni SPT unit and its technical results at Fakel	□ -To develop 0s) and power co manufacture the x he small SPT;□ - he di integrated small resentative will par B but will not be r , most of the work s:□ -the vacuu sts.□□ - the vacuu (FC tests.□□ - ed at Edward's AF w acceptance and it manufacturing an tel□□ Payment: { ent: 6 months□□ -	p, manufacture and to insumption ~100 W. enon flow controller (2). To integrate the smal SPT unit (with cathod ticipate in some of the e-engineered or disas will conducted at Fak m chamber with dia um chamber with dia inchamber suith dia 1-Dynamic pressure du B SPT test facility. For contract signing. Del did testing at Fakel 10,000 within 1 mor Deliveries: Final repo	est the laboratory model of small Emphasis will be placed upon (FC) for the small SPT; □□ - I SPT with cathode and XFC and de and XFC), test at Fakel and to be testing at Edwards AFB. Once is sembled and then reassembled. The interest is the sembled and then reassembled and then reassembled in the sembled and then reassembled. The interest is the sembled and then the semble sem
14. SUBJECT TERMS				15. NUMBER OF PAGES
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INTRODUCTION

A wide range of scientific and applied tasks can be solved by means of small and micro spacecraft. This trend is now being realized in TechSat21 program. If spacecraft mass is 100 kg then propulsion system should have a mass of ~ 10 kg and a power of about 100 W. For this purpose electric thrusters of various types can be used. The use of small stationary plasma thruster (SPT) is promising. Demonstration test results of such a thruster are adduced in this report.

After the first stage of the demonstration test [2] the anode unit, cathode unit and flow control unit were assembled as a monoblock unit – a small SPT. The purpose of testing was to check the small SPT for compliance with the 1590p project requirements:

- full power 100 W;
- anode efficiency (without taking into account the cathode flow rate) > 20%;
- specific impulse (without taking into account the cathode flow rate) >800 s.

STRUCTURE DESCRIPTION

Overall and mounting dimensions of the small SPT are shown in Fig. 1. The thruster unit consists of the SPT-25 anode unit, the BNK-0.5 heaterless cathode, the FCU-0.5 flow control unit and a common plate. Mass of the thruster unit is 740g. General view of the small SPT laboratory model is shown in Fig. 2.

Electropneumatic circuit diagram of the test is shown in Fig. 3. To initiate the discharge in the heaterless cathode the ignition unit laboratory model was used.

TEST RESULTS

Fig. 4 shows the view of operating thruster.

The volt-ampere characteristic at the 3 levels of flow rate are adduced in Table 1. Operating time at final test was 9 h.

RESULTS FINALS TESTES

Volt-ampere characteristic at 3 level flow rate Table 1

	Efu	%	6,5	8,6	10,8	12,0	14,1	15,7	17,2	17,9	18,3	17,0	15,6	15,6	9,1	13,1	17,7	19,0		14,0	15,2	16,2	15,9	13,4	13,4	13,8		14,0
	Efa	%	8,0	10,5	13,1	14,6	17,0	19,0	20,7	21,4	21,9	20,3	18,6	18,6	11,0	15,7	21,1	22,6		17,2	18,5	19,8	19,3	16,2	16,1	16,6	17,0	16,8
•	Ispu	S	367	447	524	572	641	709	770	818	867	867	863	891	460	268	786	887		647	701	754	781	745	772	808	843	870
	Ispa	s	428	522	611	899	748	828	868	955	1011	1011	1006	1039	233	<i>L</i> 69	617	1035		755	817	880	911	698	<u>8</u>	942	984	1015
l able 1	īŽ	W	61,8	69,5	75,8	81,2	87,0	95,4	102,8	111,8	122,7	131,6 1011	142,3	151,8	77,3	6'06	115,8	137,6		80,4	87,3	94,4	103,4	111,6	120,4	127,8	135,5	145,6
	Na	W	58,7	66,4	72,8	78,2	84,0	92,4	8,66	108,8	119,6	128,6	139,3	148,8	74,8	88,4	113,3	135,1		76,4	83,3	90,4	99,4	107,6	116,4	123,8	131,5	141,6
Voit-ampere characteristic at 3 level now rate	Pv	Топ	4,7E-05	4,7E-05	4,7E-05	4,7E-05	4,7E-05	0,62 4,7E-05	4,7E-05	4,7E-05	0,62 4,7E-05	0,62 4,7E-05	4,7E-05	0,62 4,7E-05	4,9E-05	4,9E-05	13,5 0,69 4,9E-05	0,69 4,9E-05		16,7 0,56 4,3E-05	4,3E-05	0,56 4,3E-05	0,56 4,3E-05	4,3E-05	4,3E-05	0,56 4,3E-05 123,8	0,56 4,3E-05 131,5	0,56 4,3E-05
leve	G	mg/s	0,62	0,62	0,62	0,62	0,62	0,62	0,62	0,62	0,62		0,62	0,62	69'0	69'0	69,0			0,56	0,56			0,56	95,0			
c at o	Ucg	Λ	13,9	13,9	14,1	14,6	14,9	15,1	15,2	14,9	14,5	13,9	13,6	13,3	12,5			13,2		16,7	16,4	16,6	15,9	15,4	15,2	15,0	14,8	14,5
FIST	Utt	Λ	0,97	0,97	0,97	0,97	0,97	0,97	0,97	76,0	0,97	0,97	0,97	0,97	0,84	0,84	0,84	0,84		1,17	1,17	1,17	1,17	1,17	1,17	1,17	1,17	1,17
racte	Itt	A	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,39	2,11	2,11	2,11	2,11		2,82	2,82	2,82	2,82	2,82	2,82	2,82	2,82	2,82
re cna	PFCU	kPa	110	110	110	110	110	110	110	110	110	110	110	110	110	110	110	110	٠	110	110	110	110	110	110	110	110	110
ampei	Uc	>	0,605	0,605	1,30 0,600	0,600	1,30 0,600	1,30 0,595	0,595	909'0	0,610	0,610	0,615	0,615	0,630	0,625	0,625	0,630		1,30 0,640	0,635	0,640	0,645	0,645	0,645	0,645	0,650	0,650
-110 A	lc	A	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30		1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30	1,30
	PΩ~	>	3,60	3,00	2,65	2,65	3,00	4,00	4,20	4,40	5,80	30,00	36,00	5,42 38,00	2,75	2,70	4,70	4,20		2,00	3,20	4,80	22,00	34,00	38,00	39,00	40,00	41,00
	H	ZII	2,23	2,72	3,19	3,48	3,9	4,32	4,68	4,98	5,27	5,27	5,25	5,42	3,11		5,32	6,01		3,56	3,85	4,14	4,29	4,1	4,24	4,44	4,64	4,78
	PI	A	0,630	0,650	110 0,655	0,645	0,645	0,650,	0,660	0,675	0,695	0,710	0,725	0,740	0,740	0,730	0,750	0,790		0,540	0,550	0,560	0,580	065'0	0,605	0,615	0,625	0,640
	PA	>	92	101	110	120	129	141	150	160	171	180	191	200	100	120	150	170		140	150	160	170	181	191	200	209	220
	Z		-	2	3	4	5	9	7	∞	6	10	=	12	-	7	3	4		1	7	m	4	5	9	7	∞	6

List of parameters adduced in Table 1: Ud - discharge voltage; Id- discharge current; F- thrust; ~Ud - discharge voltage oscillation; Ic - magnetic coil current; Uc - coil voltage; P_{FCU} - pressure at the FCU inlet; Itt - thermothrottle current; Utt - thermothrottle voltage; Ucg - cathode-to-ground voltage; G - total Xenon flow rate; Pv - pressure in the vacuum chamber (by air); Na - anode unit power; Nu - total power of the SPT unit; Ispa - specific impulse without taking into account the cathode flow rate; Ispu - specific impulse; Efa - thruster efficiency; Efu - total efficiency of the thruster unit.

Ispa, Ispu Efa, Efu, Na and Nu parameters were calculated in accordance with the equations adduced in the report [2]. Main parameters have agreed with ones obtained previously.

CONCLUSION

- 1. The scheduled work has been executed in full.
- 2. Small SPT parameters comply with the requirements specified.
- 3. An ignition unit is supplied as part of small SPT. The small SPT is prepared for demonstration test at Customer's facilities.

REFERENCES

- 1. R.A. Spores, M. Bircan,. The USAF Electric Propulsion Program. IEPC-99-009.
- Presentation and demonstration test of flow control system and the cathode for the small SPT.
 Technical report. (Project 1590p)

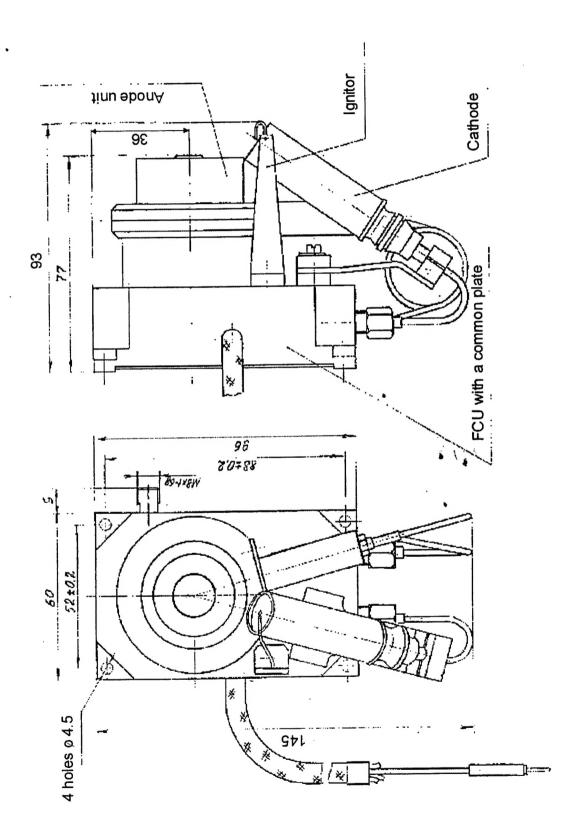


Fig. 1 Overall dimensions of small SPT

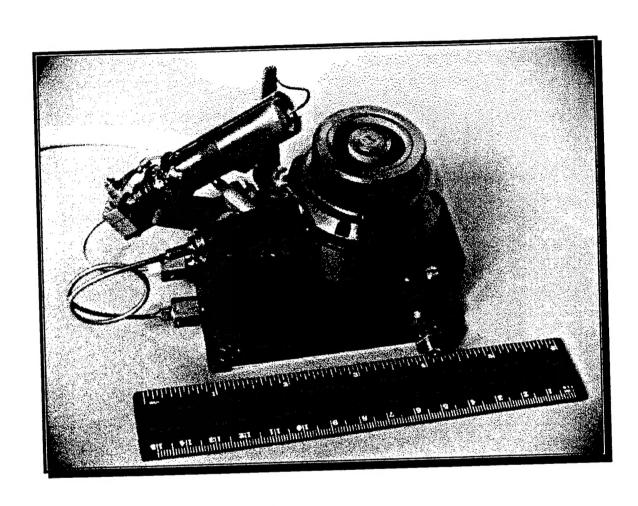


Fig. 2 Small SPT general view

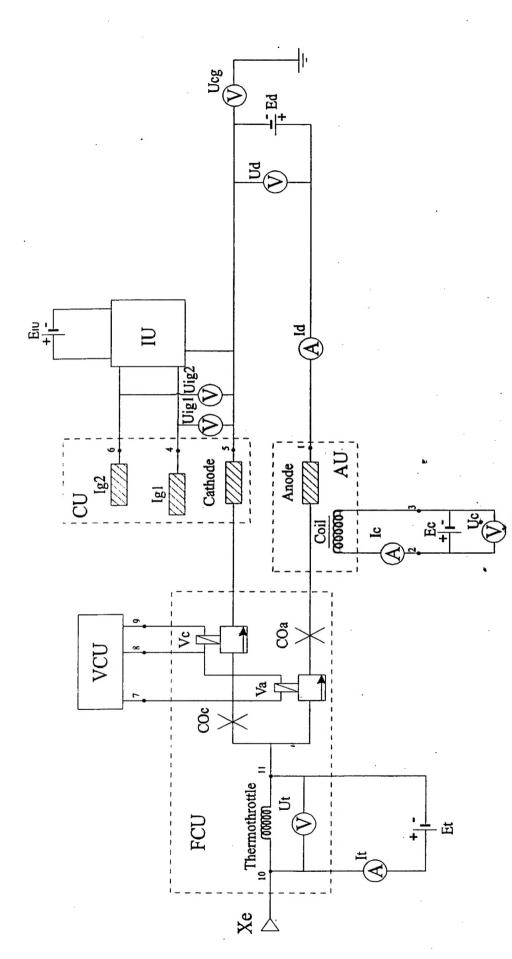


Fig.3.Pneumoelectric circuit diagram of the small SPT-25 FCU - flow control unit, CU - cathode unit, VCU - valve control unit, IU - ignition unit, AU - anode unit

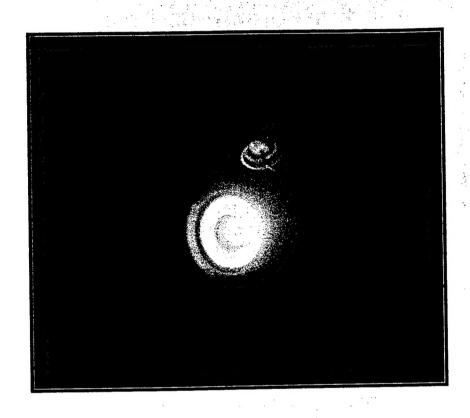


Fig. 4 View of operating thruster

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ISTC PROJECT COMPLETION STATEMENT

#1590p

TITLE OF THE PROJECT:
Development of Small SPT Demo Model

EXECUTIVE DIRECTOR'S STATEMENT:

The aforementioned ISTC project is completed. The consolidated Cost Statement has been audited and the overhead payment has been released. Please find enclosed a copy of:

- 1. Summary of the Final Project Report;
- 2. Evaluation by the Secretariat of the Final Project Report;
- 3. Consolidated Cost Statement.

Michael Kröning, Executive Director

Date: >> (1,00)

EVALUATION FORM FOR FINAL REPORTS OF ISTC PROJECTS

	Project Attributes
Project Number	1590p
Project Title	Development of Small SPT Demo Model
Leading Institute	Experimental Design Bureau Fakel, Kaliningrad, Kaliningrad Reg., Russia
Project Manager	Arkhipov Boris
O.C.D.	01 November 1999
Duration	4 months
Total Budget	\$ 35,000
Funding Parties	United States Air Force / The European Office of Aerospace Research and Development, London, UK
ISTC Project Manager	S. Karabashev
ISTC Deputy Executive	S. Zykov
Director	
	Major Technical Accomplishment

I. Accomplishment of all tasks of the Work Plan

The main results of the project can be summarized as follows:

- 1) A small SPT(stationary plasma thruster) demo model was developed, manufactured and tested.
- 2) The small SPT demo model consists of:
- SPT-25 laboratory model of the thruster;
- Cathode;
- Flow control unit (FCU).
- 3) Each part of the small SPT demo model has successfully undergone autonomous tests and the declared performance level has been confirmed.
- 4) The assembled small SPT demo model has undergone demonstration tests and showed compliance with specified requirements, as follows:
 - thrust efficiency $\eta_{\tau} \ge 0.2$, calculated without cathode mass flow rate;
 - specific impulse (800-1000)s, calculated without cathode mass flow rate;
 - power ~100W.
- 5) The small SPT demo model was delivered to the Partner for demonstration tests.

II. Published Papers, Presentations, Trips (Conferences, Meetings).

Data obtained in the project have not been published

III. Collaboration between CIS Institutes.

The project was performed by one Institution.

IV. Partnership with Foreign Institutes.

This is a partner project.

V. Technology Implementation Plan.

This is a partner project.

Financial Monitoring

The audited Consolidated Cost Statement is attached.

Overhead payment has been made in accordance with the results of the final audit.

Equipment and Materials

In total \$990.39 was spent on materials

Technical Monitoring

I. On-site monitoring was performed at the Experimental Design Bureau Fakel. No major deviations from the Work Plan or ISTC project management regulations were identified.

Other Comments

S. Karabashev

Senior Project Manager

Date: 20.11.007.

S, Zykov

Deputy Executive Director

Date: 21.11.00

I S T C M H T LI		al Science and Technolo endor Payment Request	
Invoice Number:	IN-P - 1590	ADM or Project Num.	P 1590
For Institute/City	CDB "Fakel" , Kalining	rad	QR
Payee Name and Address:			
Contract/Invoice Num.			
Details	Overhead Disbursemen	nt	
Terms		(Single P	MT/ Multy stage PMT
Amount to	be Paid and Distributed	\$ 407.95	Currency
Requested by Allotment	Manager	4	Date
Title/Name	Dr. Michael Kroenin	g, ED	-3, 11.02 V

Accounting Information (to be filled by Finance Office Staff) **Budget Control Officer** Title/Name Date Received Invoice Number: (AP-Invoice- Document Type = Invoice; Purchase=Bank ACCT) ACCPAC input Job-Phase- Category Amount Account Num Amount P1590 - ZY - 7AA1 For Admin Only 407.95 Exchange Rate: 407.95 TOTAL \$ Inputted by ; Batch Num Input Date **Payment Processing** CONV/RUB Bank Code (Check one) BTC CONV/DOLL Amount USD/RBL Date DETAILS PMT Approved: Initial Date **BATCH Num** Initial (AP-Payment- Document Type =Manual Check) ACCPAC input Inputted by Input Date ; Batch Num Check Num

Project

1590

FINAL FINANCIAL REPORT BY AUDIT. ADJUSTMENTS.

Reporting period:

November, 01, 1999 - April, 30, 2000

QR

		Cost Category	Accumulate Recipient		Accumulated C	Cost per Audit	Adjustm	ents
		Cost Gategory	(1)	(2)	(1)	(2)	(1)	(2)
1	_	GRANT PAYMENTS						
		Category - I		21,500.00		21,500.00		0.00
		Category - II		9,400.00	-	9,400.00	-	0.00
		Category - III			_		L	0.00
		Category - IV		920.00		920.00		0.00
		Total Grant Payments		31,820.00		31,820.00		0.00
2	-	Equipment:						
	1	Modifications						
	2	Capital Equipment						
	3	Non-Capital Equipment						
	4	Leased Equipment						
	5	Maint & Repair						
		Including VAT						
	_	Total Equipment						: -
3		MATERIALS						0.00
	1	Materials	470.04	990.39	470.04	990.39	0.00	0.00
	2	Supplies						
	3	Safety Devices						
	4	Other						
	_	Including VAT						
	_	Total Materials	470.04	990.39	470.04	990.39	0.00	0.00
4		BANK FEES	3.35	135.50	3.36	141.35	0.01	5.85
5	_	OTHER DIRECT COSTS						
	1	Technological Energy						
1	2	D /D blishing						
	3	0						
	4	Admin. Supplies						
	5							
		Including VAT						
		Total ODC					Mag. 1	
6	_	TRAVEL\PER DIEM						
٥		Local-Russia-CIS	429.60		432.66		3.06	
		Outside CIS						
		Total Travel & Per Diem	429.60		432.66		3.06	
7		Exchange rates gains\losses	-7.65	6.31	-10.72	6.31	-3.07	0.00
8		Overhead		0.00)	738.66		738.66
F	_	TOTAL	895.34	32,952.20	895.34	33,696.71	0.00	744.5
	_	TOTAL VAT INCLUDED						
Г		GRANDTOTAL		33,847.54	1	34,592.05		744.51

Remarks: *(1) - Cash flow through Recipient Account

ISTC Auditor

Timur Timerbaev



3 11 JHT 2000

^{** (2) -} Cash flow through ISTC

CDB Fakel, Kaliningrad

Project

1590

FINAL FINANCIAL REPORT BY AUDIT. RESIDUALS

Reporting period:

November, 01, 1999 - April, 30, 2000

QR

	Cost Category		Budget ¹	TOTAL	Accumula	ted Cost	Funds Residuals		
		-	(1)	(2)	(1)	(2)	(1)	(2)	
1	GRANT PAYME	NTS							
	Category - I			21,500.00	L	21,500.00		0.00	
	Category - II			9,000.00		9,400.00	L	-400.00	
	Category - III				L		Ĺ		
	Category - IV			720.00		920.00		-200.00	
	Total Grant Pay	ments		31,220.00		31,820.00		-600.00	
2	Equipment:								
	1 Modifications								
	2 Capital Equipme	ent							
.;	3 Non-Capital Equ	uipment							
,	4 Leased Equipm	ent							
	5 Maint & Repair								
	Including VAT								
-	Total Equipm	ent							
3	MATERIALS								
	1 Materials		0.00	1,000.00	470.04	990.39	-470.04	9.61	
	2 Supplies								
	3 Safety Devices								
	4 Other								
_	Including VAT								
-	Total Mate	rials	0.00	1,000.00	470.04	990.39	-470.04	9.61	
4	BANK FEES		4.50	157.50	3.36	141.35	1.14	16.15	
5	OTHER DIRECT	COSTS							
	1 Technological E	nergy							
	2 Reports/Publish	ing	500.00		0.00		500.00		
	3 Communication	s							
	4 Admin. Supplies	5							
	5 Other								
	Including VAT								
	Tota	al ODC	500.00		0.00		500.00	46.4	
6	TRAVEL\PER D	IEM							
	Local-Russia-C		400.00		432.66		-32.66		
	Outside CIS								
	Total Travel 8	Per Diem	400.00		432.66		-32.66		
7	Exchange rates	gains\losses			-10.72	6.31	10.72	-6.31	
8	Overhead			0.00		738.66		-738.66	
	TOTAL		904.50	32,377.50	895.34	33,696.71	9.16	-1,319.21	
	TOTAL VAT INC	LUDED							
	GRANDTOTAL			33,282.00		34,592.05		-1,310.05	

Remarks: *(1) - Cash flow through Recipient Account

** (2) - Cash flow through ISTC

ISTC Auditor Timur Timerbaev The Thusupsall

引作 洲 柳原

Наименование Рабочей Бумаги	РАСЧЕТ СУММЫ НАКЛАДНЫХ РАСХОДОВ К ВЫПЛАТЕ
Проверку проводил: Тимербаев Т.Ф.	
Дата: 22 июня 2000 г.	

Проскт

1590 P

Институт

ОКБ "Факел", гор. Калининград

QR

	Сумма	Формула	Комментарий	
Всего произведено затрат (1)	895.34	k		
Всего произведено затрат (2)	33,696.71	1	Включая: сумму накладных расходов в сумме \$ 730.00 + остаток на счете \$ 8.66 (см. ратыснения пиже)	
Итого ФАКТИЧЕСКИХ прямых затрат по				
проекту	34,592.05	m=k+l		
Итого ПЛАНОВЫХ прямых затрат по				
проекту	32,562.00	n		

Если фактических прямых затрат больше, чем	плановых		m>n
Общая стоимость проекта	35,000.00	0	
Затраты (1) подлежащие компенсации	0.00	t	
Остаток средств на счете	0.00	d	
Начисленные Накладные	0.00	u=o-m	
Накладные расходы + компенсация по прямым затратам к выплате	0.00	p=u+t-d	

Если фактические прямые затраты меньше и	ли равны, плано	вым т<=п	
пановые накладные расходы	2,438.00	r	
Затраты (1) подлежащие компенсации	895.34	t	
	9.77	a	Остаток на банковском счете (российские рубли) составил RUB 247.67. Курс МНТЦ на
Остаток средств на счете	8.66	d	01.04.2000 составляет 28.60 руб. за \$ 1.
Начисленные Накладные *	407.95	u=r	
Накладные расходы + компенсация по			, ,
прямым затратам к выплате	1,294.63	s=u+t-d	
			1
			Сэкономленные средства перезачтены на UCC
Общая экономия средств по проекту **		x=0-m-u	(130110) соответствующих сторон

Использован следующий расчет: 35,000.00-33,862.05-730.00 = 407.95, где 35,000.00 общая стоимость проекта, 33,862.05 расходы проекта за весь период. 730.00 - часть суммы накладных расходов, уже использованных ОКБ "Факсл" для отправки экспериментальных образцов коллаборатору проекта 1590.

The Tunipowb

Реально экономии по проекту нет: формула ISTC F-A8 не учитывает \$ 730.00, а также \$ 8.66 (см. ссылку *) **